

PATENT

5

INVESTIGATION OF DESTROYED ASSEMBLIES AND

IDENTIFICATION OF COMPONENTS THEREOF

10

Background of the Invention

This invention pertains to methods of investigating
15 accidents or intentional acts leading to the destruction of
assemblies. In particular, this invention pertains to the
identification of components that previously formed part of an
assembled object and to a method of investigating the cause of
the destruction of the assembly.

20 In situations such as an investigation of the cause of an
aircraft crash, a detailed reconstruction of the events
leading up to the destruction of the aircraft is often
conducted. In addition to recorded data and witness
knowledge, the recovered components of the aircraft are often

useful in identifying the cause of the crash and, in the cases where no recorded data and witness knowledge is available, scrutiny of such components may be the only means of conducting an investigation.

5 The analysis of the components of a destroyed assembly, such as an aircraft, typically begins by locating and collected recoverable components. In some situations, aircraft crashes in particular, locating the components can be a difficult task due to the fact that such components are
10 often spread out over a very large area. For example, following the destruction of the Space Shuttle Columbia, more than 8,000 separate components of the shuttle were located scattered over hundreds of square miles. As a result of the difficulties associated with locating such components,
15 invariably many components are never located.

For those components located, the next step in an investigation often is to identify each particular component so as to be able to associate the component with a portion of the assembly, as the component was associate therewith prior
20 to the assembly's destruction, and with the remainder of the located components. Typically this involves visual inspection of each component by persons highly familiar with various components of the original assembly who, by such inspection, can identify at least some of the components. In other

situations, drawings of the components of the assembly that were created prior to the assembly's destruction are used to aid those attempting to identify the located components. In yet other situations, comparisons are made to physical 5 components of similar undestroyed assemblies. Thus, the task of identifying the located components can, in itself, be tedious and time consuming.

The next step in an investigation typically comprises physically rigging the located components onto a structure in 10 an effort to reconstruct the assembly, at least to the extent possible. When rigging the components, the components are positioned relative to each other in manner emulating their respective original relative positions prior to the destruction of the assembly. The reconstructed portion of the 15 assembly provides investigators with valuable information by allowing them to visualize the various components, and thereby speculate on the possible cause of the assembly's destruction. However, the rigging of the components can be difficult and expensive due to the fact that unique structure is often 20 required to support the components in their relative positions. In some cases, the cost of the structure supporting the rigging, alone, can be extensive. Moreover, rigging often limits the mobility of the components, thereby possibly making it difficult for investigators to obtain

desired information from the components. Furthermore, rigging also generally requires a dedicated facility, which can be costly and may reduce or limit the ability of investigators to obtain access to beneficial data. Yet 5 further, the rigging of a component may require physical modification to the component, and may thereby potentially compromise the evidentiary data provided by the component.

In view of the foregoing, it should be appreciated that the reconstruction of a destroyed assembly can be a valuable 10 tool in the investigation of the cause of destruction of such assembly. However, it should also be appreciated that, while advantageous and worthwhile, such reconstruction typically involves numerous steps that are laborious, that require expertise, that consume large periods of time, and that are 15 expensive.

Summary of the Invention

The present invention eliminates many of the disadvantages associated with investigations of destroyed 20 assemblies by eliminating the need to perform many of the steps previously required during the reconstruction of such assemblies. Despite the elimination of such steps, the invention nonetheless provides the benefits associated with such prior art reconstructions. In general, the invention

provides a means for identifying component parts quickly and relatively easily and provides a means for digitally rigging the component parts in three-dimensional virtual space, thereby minimizing any need to physical rig the component

5 parts. Yet further, the digital or electronic rigging of the reconstructed assembly enhances the ability to investigate the component parts. In fact, at the time this specification was being prepared, various aspects of the invention were being utilized in connection with the investigation of the cause of

10 destruction of the Space Shuttle Columbia and had proven to be superior in some applications while being complementary to investigation methods developed prior to the invention. These and other aspects of the invention are discussed hereafter.

A first embodiment of a method of practicing the

15 invention pertains to a method of analyzing a cause of destruction of an assembly. This method generally comprises obtaining first and second components that formed part of the assembly when the assembly was in the assembled condition, producing an electronic representation of a three-dimensional

20 surface contour of at least part of each of the first and second components, producing an electronic representation of a portion of the assembly in three-dimensional virtual space, and analyzing the cause of destruction of the assembly via the representation of the portion of the assembly. The first and

second components have been separated from the assembly and
from each other as a result of the assembly being destroyed by
the cause of destruction and had an original relative position
between each other prior to the assembly being destroyed. The
5 electronic representations of a three-dimensional surface
contours of the first and second components are produced by
obtaining measurements of each of the first and second
components using a surface contour scanning device. The
electronic representation of the portion of the assembly
10 incorporates the representations of the three-dimensional
surface contours of the first and second components with the
surface contours positioned relative to each other in the
virtual space in a manner based on the original relative
position of the first and second components.

15 A second embodiment of a method of practicing the
invention pertains to a method of identifying a component of a
dismantled assembly. Prior to being dismantled, the assembly
was in an assembled condition. This method generally
comprises obtaining an electronic database of properties of a
20 plurality of components that are presumed to have been part of
the assembly when the assembly was in the assembled condition,
retrieving a physical component that has been dismantled and
separated from the assembly, identifying a property of the
physical component by obtaining measurements of the physical

component, and correlating the physical component to one of the plurality of components by matching the property of the physical component to one of the properties of the plurality of components of the database. The properties of the 5 plurality of components in the database are based at least in part on geometry that the plurality of components are presumed to have had when the assembly was in the assembled condition. The property of the physical component is identified by obtaining measurements of the physical component and by 10 producing an electronic representation of a three-dimensional surface contour of a portion of the physical component based on the measurements.

While the principal advantages and features of the invention have been described above, a more complete and 15 thorough understanding of the invention may be obtained by referring to the drawing and the detailed description of the preferred embodiment which follow.

Brief Description of the Drawings

20 Figure 1 is flow chart showing steps of the preferred method of practicing the invention.

Detailed Description of the Preferred Embodiment of the
Invention

A flow chart of the preferred method of practicing the invention is shown in Figure 1 and is particularly adapted for 5 the investigation of a destroyed aircraft. For purposes of describing the preferred method, it should be appreciated that such an aircraft is presumed to have been disassembled into a plurality of component parts as a result of an unknown cause of destruction. Furthermore, at least some of the components 10 of the originally assembled aircraft are presumed to be recoverable.

In general, the preferred method begins by obtaining components of a disassembled assembly. Preferably, these components are transported to a central location from the 15 locations where they were first located after being separated from the assembly. However, because the preferred method of practicing the invention does not necessarily require physical rigging of the components in an effort to partially reconstruct the assembly, relocation of the components is not 20 necessarily required.

After a component has been obtained, an initial scan of the component is performed. The initial scan preferably comprises obtaining measurements of at least a portion of the component's surface contour and creating a digital or

electronic representation of at least a portion of the component's surface contour in three-dimensional virtual space. This is preferably done using a photographic three-dimensional scanner and software, a laser three-dimensional scanner, or other three-dimensional scanner commercially available for creating digital three-dimensional representations of objects quickly and without directly contacting the component being scanned. However and alternatively, other scanning methods or devices, such as coordinate measuring machines and even hand measurements, could be used to scan an object in a manner to achieve the ultimate result of producing a digital three-dimensional representation of the component's surface contour.

Optionally or in conjunction with the initial surface scan of the component, a digital image of the component's surface can be obtained. Preferably, the digital images are obtained via a digital camera or color three-dimensional scanner but can also be obtained by converting non-digital images such as film photographs into digital form or using other well known techniques. The digital image can be a planar two-dimensional image, but is preferably a three-dimensional color image that can be texture mapped to the digital three-dimensional representation of the component's surface contour.

At this stage in the preferred method, an attempt to identify the component is made. To this end, the component or a rendering of its represented surface contour is preferably initially visually examined in an effort to determine if the 5 identity of the component is readily apparent. For example, if the component happens to be a severed, but otherwise undamaged, wing tip of a destroyed aircraft, its identity as such would presumably be readily apparent.

If on the other hand, the identity of the component is 10 not readily ascertainable, the representation of the surface contour of the component is preferably electronically compared to representations of surface contours of the components presumed to have together constituted the assembly. For clarity, these latter components are herein referred to as the 15 virtual components of the assembly and the scanned component is herein referred to as the physical component. The representations of surface contours of the virtual components are preferably obtained from a preexisting computer aided design (CAD) solid or surface model of the assembly. 20 Alternatively, a CAD model of the assembly in its assembled condition could be created specifically for practicing the invention by scanning the components in a similar assembly that has not been destroyed or by converting two dimensional production drawings of such components into three-dimensional

surface or solid CAD models. However, in the case of most modern aircraft, as well as automobiles, pre-existing solid CAD models often exist.

In performing the surface contour comparison, a portion 5 of the represented surface contour of the physical component is preferably selected for the comparison. Preferably, the portion selected is presumed to be unique to the physical component and to be in an undeformed condition. The portion could be merely a two-dimensional surface contour curve 10 extrapolated from the three-dimensional represented surface contour of the physical component. The comparison is preferably carried out via an automated computer routine to minimize human labor requirements but could also be performed manually using CAD software. When and if a match between the 15 portion of the surface contour of the physical component and the represented surface of one of the virtual components occurs, the component is correlated to the particular virtual component.

Alternatively or in conjunction with the comparison of 20 the represented surface contour of the physical component with the surface contours of the virtual components presumed to have together constituted the assembly, any fracture edge of the physical component that is included in the represented surface contour of the physical component can be compared to

other represented fracture edges of other represented physical components in an effort to identify matches of corresponding fracture edges. Thus, it should be appreciated that the represented fracture edges can be compared to each 5 other to identify the correlation of between a physical component and one or more of the virtual components indirectly.

As yet another alternative, markings and color patterns obtained from the digital images taken of the physical 10 components can be compared to each other, preferably in an electronically automated manner, in an effort to correlate any particular representation of a physical component with that of another. This is particularly useful in situations where the digital images include markings such as those 15 created by exposure to plasma flow or relatively unique abrasion marks.

If the surface contour and/or digital image comparisons were unsuccessful, a detailed scan of the physical component is then preferably performed. The detailed scan is preferably 20 performed using the same technique as was used in performing the initial scan. However, the detail scan involves a complete surface scan of the entire surface contour of the physical component so as to create a detailed digital representation of the entire surface contour of the physical

component. From this detailed digital representation, it should be appreciated that the volume of the physical component and other information such as the principle moment of inertias of the physical component can be readily

5 determined using CAD software or other particularly adapted computer software.

At this stage in the preferred method, the surface contour comparison discussed above can be repeated using the detailed digital representation of the surface contour the

10 physical component in place of the initial digital representation. It should be appreciated that a surface comparison using the detailed representation of the surface contour of a physical component may be able to correlate the physical component with a virtual component, despite the

15 failure of such comparison using the initial surface scan.

If, at this stage, a correlation has not been achieved, the volume of the physical component obtained from its detailed representation is compared to the volume properties of the virtual components in the database in an effort to

20 identify the component. It should be appreciated that the surface representations of the virtual components allows the volume of each virtual component and other information such as the principle moment of inertias of each virtual component to be determinable. As was the case with the surface comparison,

the volume comparison is preferably carried out via an automated computer routine to minimize human labor requirements but could also be performed manually using CAD software. Likewise, when and if a match between the volume of 5 the physical component and the volume of one of the virtual components occurs, the component is correlated to the particular virtual component.

If the volume comparison is unsuccessful, a comparison of the principle moment of inertia of the physical component, as 10 determined from the digital representation of physical component, to the principle moment of inertias of the virtual components can then be performed. Once again, the moment of inertia comparison is preferably carried out via an automated computer routine to minimize human labor requirements but 15 could also be performed manually using CAD software. It should be appreciated that the moment of inertia of a component, assuming the component is not severed or deformed, acts as a unique fingerprint of the particular component for comparison purposes. Thus, in situations involving many 20 similar but unique components existing in essentially their original shape, such as with the heat shield tiles from the Space Shuttle Columbia, matching by moment of inertia comparison can be highly successful. This being said, when and if a match between the moment of inertia of the physical

component and the moment of inertia of one of the virtual components occurs, the component is correlated to the particular virtual component.

Aside from the volume and inertia properties of the 5 components obtained from the detailed surface scans, such surface scans can be utilized to fabricate detailed physical models of the components using digital fabrication techniques such as stereo lithography, selective laser sintering, and fused deposition modeling. Such physical models of the 10 components are beneficial in several ways. First, producing a physical model of a component allows investigators to hold or touch a tangible model of the component and to possibly identify the component based thereon. It should be appreciated that physical handling of the actual physical 15 components may be difficult or dangerous, as is the case with many of the Space Shuttle Columbia components which emit dangerous levels of radioactivity. Additionally, the digitally fabricated models can be produced at locations that are convenient to the investigators, thereby eliminating the 20 need to transport the actual physical components or the investigators themselves. Yet further, the digitally fabricated models of the components generally weigh much less than that the actual physical components and are thereby easier to manipulate during the investigation process.

Finally, the digitally fabricated models of the components may be utilized in a physical rigging of the assembly, or a physical model of the assembly, as is discussed later in this description.

5 For those physical components that have been successfully identified, the original positions of the physical components relative to each other, i.e., prior to the destruction of the assembly, are ascertained. This is done via the database or CAD model of the virtual components which inherently contains
10 information indicative of the positions of the virtual components relative to each other. Thus, the original relative position between any two physical components is assumed to be the relative position between the two virtual components that correlate respectively to the physical
15 components.

 The preferred method further comprises creating a virtual rigging or virtual reassembly of at least a portion of the assembly. This is done by positioning each of the representations of the surface contours of the physical
20 components in virtual three-dimensional space based on their respective original relative positions. For those physical components that have only been initially scanned, a detailed scan of the component can be performed if it is determined

that greater detail of the component in the virtual rigging is warranted.

As an option, texture-mapping can be added to the virtual rigging. This can be done for some or all of the components, 5 as desired. For any component, the texture-mapping is performed by obtaining digital images of the visual appearance of some or all of the actual surface contour of the physical component, as discussed above. Once the desire digital images are obtained, they are texture-mapped to the digitally 10 represented surface contour of the physical component using conventional texture-mapping techniques.

Just as conventional physical rigging of physical components allows investigators to visualize the various components of a destroyed assembly in an organized manner, so 15 too does the virtual rigging. Moreover, with texture-mapping included in the virtual rigging, investigators are able to analyze the texture-mapped surface to identify such things as material degradation and gas/plasma flow residue on various components. Thus, the virtual rigging provides all of the 20 advantages of a physical rigging without requiring an investigator to visit a specific site or view a specific component and without requiring expensive rigging structure and the many man hours of labor associated therewith.

Beyond the advantages described above, the preferred method also provides other advantages. For example, if desired, finite element analysis can be performed on desired components in an effort to determine the forces that caused 5 their deformation or for other reasons. This can be done simply by using the digital representation of the surface contour of the physical component and creating a finite element model directly therefrom using conventional finite element modeling software.

10 Yet another advantage includes the ability to estimate aerodynamic properties of any of the physical components using computational fluid dynamics or other well known methods based on the represented surface contours of the components. Using these aerodynamic properties in combination with other known 15 data, such as the presumed location of a component when it came to rest after being separated from the assembly (typically presumed to be the location where or component was recovered or where an impact is evident), allows investigators to estimate the trajectory of the component between the time 20 it presumably separated from the assembly and the time at which it presumably came to rest. Alternatively, these aerodynamic properties allow investigators to estimate the time and location that a component presumably separated from

the assembly or from some other portion of the assembly that had already separated from the main assembly.

The trajectory estimates are useful not only to the investigation of the obtained components, but also can be utilized to estimate the trajectory of components that have not been located. It should be appreciated that aerodynamic properties of unfound components can be calculated using the surface contour data of the virtual components that presumably correlate to the unfound components. Thus, based on assumptions gleaned from the estimated trajectories of components positioned adjacent to the unfound components prior to the destruction of the assembly, the trajectory of the unfound components can be estimated. This allows investigators to focus their search for unfound components to specific areas estimated to be the rest locations of such unfound components. Thus, the trajectory estimates can increase the number of components ultimately recovered from destroyed assembly as well as provide insight into the cause of the destruction of the assembly.

It is important to appreciate that the physical components recovered from the destroyed assembly are often combinations of two or more portions of correlating virtual components and are often deformed from their original geometry. Thus, it should be appreciated that when practicing

the preferred method of the invention, investigators
preferably attempt to identify specific portions of the
physical components that would likely correlate to just one of
the virtual components in the database. The surface contour
5 of the specific portions of the physical components are then
preferably digitally represented independent of each other
during the surface contour comparison. Additionally, the
physical component can alternatively be disassembled into a
plurality of physical components that are each believed to
10 correlate directly to a single one of the virtual components.
Moreover, various steps can be performed to increase the
probability of correlating obviously deformed or not fully
intact physical components to the virtual components. Such
steps include altering the represented virtual surface contour
15 obtained from an obviously deformed physical component in a
manner such that the represented surface contour reflects the
presumed original or undefomed surface contour of the
physical component. This is most beneficial in performing the
moment of inertia comparison since the overall geometry of a
20 component, and not just its volume, has a direct impact on the
comparison. Similarly, if the geometry of a missing portion
of a physical component can be estimated from the portion of
the component that is intact, the estimated missing portion
can be added to the digital representation of the component's

surface contour. This is beneficial to both the volume and the moment of inertia comparisons.

The information used to produce the virtual rigging can also be utilized to provide individual part visualization
5 and/or to produce computerized visual animations. Such visualization can include the details of the texture mapping discussed above and, in the case of the animations, may utilize the trajectory estimates. Furthermore, these visualizations can facilitate the investigation of the
10 assembly or can facilitate the communication of the results and conclusions gleamed from the investigation.

As discussed earlier, the surface scans obtained from the physical components can be used to digitally produce physical models of the components. These physical models of the
15 components can be utilized to produce a physical rigging of the assembly for further investigation purposes. Such a physical rigging could include a combination of actual physical components and physical models of components, or could be entirely made up of modeled components. Due to the
20 fact that modeled components are generally lighter in weight than are the actual physical components they represent, the structure required for supporting the rigging can be simplified compared to the structure that would be required to support the actual physical components. Additionally, rigging

mounts can be added to the virtual representations of the components and can thereby be automatically created as integral portions of the modeled components during the digital fabrication process.

5 The physical rigging of the modeled components provides many of the advantages attributable to conventional rigging of the actual components, while also providing other advantages thereover. For example, rigging of the modeled components eliminates the need to modify or alter the actual physical
10 components from their original recovered condition, as is often required when mounting actual physical components to a rigging. Additionally, investigators analyzing the physical rigging of the modeled components are not exposed to hazardous material, as would be the case if hazardous actual components
15 are rigged. Furthermore, the modeled components can be scaled in size relative to the actual components to facilitate the investigation thereof. Yet further, multiple riggings in different locations could be constructed to make such riggings more accessible to investigators.

20 In view of the forgoing, many advantages of the preferred method of practicing the invention should be appreciated. However, it should be understood that all matter contained in the above description or shown in the accompanying drawing is intended to be interpreted as illustrative and not in a

limiting sense and that various modifications and variations to the preferred method may be employed without departing from the scope of the invention defined by the following claims.

For example, it should be appreciated that while the preferred

5 method of practicing the invention is specifically adapted to the investigation of destroyed aircraft, the invention is not so limited and that many if not all of the steps and aspects of the preferred method could be utilized to investigate other dismantled objects, such as collapsed or otherwise destroyed

10 land based structures and wrecked land vehicles. Moreover, not all of the steps of the preferred method of practicing the invention need to be performed, nor need to be performed in any particular order, to practice the claimed invention.

Thus, other possible variations and modifications of the

15 preferred method should be appreciated.

Furthermore, it should be understood that when introducing elements of the present invention in the claims or in the above description of the preferred embodiment of the invention, the terms "comprising," "including," and "having"

20 are intended to be open-ended and mean that there may be additional elements other than the listed elements.

Similarly, to the extent the term "portion" is used in the claims or is added by amendment, such term should be construed

as meaning some or all of the item or element that it qualifies.